



AN INVESTIGATION OF THE RATE OF FLOW OF VARIOUS GASES THROUGH CAPILLARY TUBES. BY

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An investigation of the rate of flow of various gases through capillary tubes.

by Joseph R. Branham. 1923



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RATE OF FLOW OF DIFFERENT GASES THROUGH CAPILLARY TUBES When a gas is flowing through a tube there is a certain resistance to this flow that must be overcome by a pressure difference between the ends of the tube. For a given rate of flow, a long tube requires a greater pressure difference than a short one, and a tube of small diameter, a greater pressure difference than a tube of large diameter. The rate of flow is therefore dependent upon the length of the tube, its diameter, and the pressure difference that exists between its ends. For any tube, the dimensions of which are fixed, the rate of flow and the pressure difference should be functions of one another.

Advantage is taken of this fact in the construction of flow-meters. The gas is made to flow through a tube of small diameter; the pressure difference of its ends is measured, and from this measured difference the rate of flow is deduced.

There are four types of flow-meters in general use. They are the resistance tube flow-meter, the orifice flow-meter, the Venturi flow-meter and the Pitot tube flow-meter. Only the first type is dealt with in this article.

This type consists, essentially, of a capillary tube set into the main gas line and a device whereby the pressure difference between the ends of this tube can be measured.

The usual form of the resistance tube flow-meter is shown in figure (1).







PURPOSE

On account of the lack of data, in literature, which could be used as a guide in the construction of flow-meters it was decided to carry out a set of experiments to determine the rate of flow of various gases through capillary tubes. The tubes were to be of such dimensions as to be suitable for flow-meter construction.

Poiseuille" has derived a formula which gives very exact results provided the pressure gradient on the gas flowing through the tube be slight. The more nearly the pressure gradient approaches zero, the closer is the agreement between the calculated and the experimental results.

Graham's⁽²⁾work on transpiration numbers also required a small pressure gradient, which he obtained by using tubes, of which the ratio of length to diameter was greater than 4000 to 1.

Quite recently, 1918, Dr. A. F. Benton carried out a series of experiments with air flowing through capillary tubes, from which he was able to obtain data which enables one to predict closely the rate of flow of air through capillary tubes.

It is the purpose of this research to obtain experimental data on gases, other than air, which will enable us to predict the rate of flow of these gases through flow-meters and to find the relations that hold between different gases under the same conditions.

(Ganot's Physics, page 136

 (2) Chemical and Physical Researches of Thomas Graham, page 108
(3) Journal of Industrial and Engineering Chemistry, Vol. 2, page 623 (9919)



THEORETICAL DISCUSSION

Poiseuille's Law(1) is obtained upon the supposition that the there are no eddy or cross currents in the gas flowing through the tube. Under this condition of streamline flow the resistance tending to stop the flow is a measure of the gaseous viscosity alone.

Under conditions existing, where the rate of flow of gases is measured by flow-meters, there is necessarily a sudden change in the forward movement of the gas at both influx and efflux end of the capillary tube. This is bound to create turbulent regions at both ends of the tube; and if the rate is very high, or the tube very short, this turbulency may not die out but may exist throughout the intire length of the tube.

We have then three possible types of flow;(1) streamline throughout the tube; (2) streamline in the middle part of the tube with turbulency at the ends and (3), turbulency throughout the entire length of the tube.

For the first type of flow Poiseuille's law will hold but this ideal type of flow is seldom encountered in actual flow-meter measurements. For the second type of flow, we have two resistance factors, viscosity and the resistance due to the sudden change of momentum of the gas molecules, which may possibly be a function of their mass. The viscous resistance may or may not operate in the turbulent regions.

In the third type, where the entire tube is subject to turbulent flow, we may have either a single resistance arising from the change in momentum of the gas molecules, or a combinations of this resistance plus a viscous resistance.

(1) Rate $= \frac{\pi h d^4}{28 \eta l}$



NOTATION

H = difference of pressure across flow-meter (cm. H30) d= diameter of tube in mms: l= length of tube in mms. r = rate of flow corrected to 760 mms. pressure 35° C γ = coef. of viscosity 30° C (gms/cm.sec) β = density of gas (gms per cc.) v = $\frac{\gamma}{\rho}$ kinematic viscosity (cm. /sec)

DESCRIPTION OF APPARATUS AND METHODS.

The apparatus used was set up in the following order; first, gas in containers under pressure; second, calcium chloride drying tubes; third, flow-meter; fourth, wet gas meter discharging into the atmosphere. Between the gas tanks and the calcium chloride tubes was placed a vertical glass tube about 3 meters high by 5 cms. diameter which had two openings in its stoppered lower end, by means of which water could be allowed to flow in or could be drawn off. In this way the water level in this tube could be changed at will. Into this tube led a smaller tube, branching by means of a T tube, from the main gas line. This small tube terminated in an orifice about a millimeter in diameter. It reached nearly to the bottom of the larger vertical tube containing the water so that by varying the height of the water above the lower end of the small tube we could allow any excess gas, above a definite pressure, to escape into the atmosphere. By this means we could maintain, at the influx end of the flow-meter, any chosen pressure regardless of change in pressure of the gas in the containers.



We first attempted to make all determinations at constant temperature of 25° C. by means of a thermostat and a long coil of glass tubing. It was found, however, that this was of little value because of the fact that the gas underwent a rapid change of temperature when it came into contact with the walls of the flow-meter. We therefore made all determinations at room temperature, reading the temperature of the gas at the efflux end of the flow-meter by means of a thermometer placed in a branch tube. From these readings we could correct the viscosity constant when-(1) ever necessary, using Sutherland's formula. The wet gas meter was equipped with a thermometer so that we could correct the observed volumes by means of the gas laws to a standard condition, namely 760 mms. pressure, 25° C.

In actual practice the viscosity constants were assumed not to vary greatly from their value at 20° C. and in the data collected no correction has been made for this slight variation over the small range of temperature encountered.

The dimensions of the capillary tubes were determined by filling them with mercury and weighing the mercury contained in a measured length. The tubes were first tested for uniformity of bore by means of a shorter column of mercury whose length was measured in a different regions of the tube. It was found that there was some variation in bore of all the tubes used. This probably leads to an error in some cases. The average diameter was taken as correct in all calculations. The tubes used were portions of longer tubes filed and broken at right angles to their axis and the ends were not rounded by heating.

()) (Phil. Mag. 31, 1893.) or Smithsonian Physical Tables page 165.



Preliminary work, with relatively short tubes, gave rise to data which when plotted, pressure against rate, did not give curves which were regular for their entire length. These curves would be regular when the pressure and rate were below a certain limit. Beyond this limit there was apparently some factor which changed

the character of the flow.

On account of this fact we decided that we must, first of all, determine the point where the curves became irregular. This point of irregularity or inflection we will call the "break".

Examination of the data showed that the breaks were more pronounced and hence more readily located when the length of the capillary tube was great when compared to its diameter. We therefore chose three long tubes of different diameters and collected data from the behavior of gases flowing through them.

The gases used were hydrogen, carbon dioxide, ethylene, and air. These four gases have the following physical properties.

gas	mol.wt.	viscosity constant0°C.	Suther land constant	viscosity constant 20°	c.
Hz	2.016	845 × 10 ⁻⁷	75.6	888 × 10 ⁻⁷	
COz	44.00	1388 × 10 ⁻⁷	249.	1486× 10 ⁻⁷	
C _z H ₄	28.04	990 × 10 ⁻⁷	249.	1060×10^{-7}	
Air		1731 × 10 ⁻⁷	112.	1819 × 10 ⁻⁷	

These gases were chosen because their volumes could be measured in a wet gas meter and because in them were found all possible combinations of high and low density, and high and low viscosities.

According to Benton, who uses a formula derived by Osborne Reynolds; r_c ; the critical rate, the point on the pressure-rate diagram where the plotted curve undergoes a sudden change of



direction is equal to ten dv. (10 dv.). That is to say, that the point of change is a function of the diameter or linear velocity and the viscosity and the density of the gas. Substitution tuting the values obtained from our data we get results which seem to be in fair agreement with this formula. This is shown as follows.



	ATD		Page 8 (B)
	AIR		
Tube	Lif. min.	lit.min.	
C	.001185 2.385	$2.52 r_{-} = 1$	1. 62 d
		- C	
N	r. 17.374 .0001819		
	.001185 2.665	2.72 $r_c =$	1.57 d
P 1	$r \frac{7.703}{.001185} 1.18$	Diameter too smal	1 to get
		limit	in pressure
	CO		
Tube	z lit.n	nin lit.min	
0	$r = \frac{15.53 .0001486}{.0018}$ cal	c. found 82 1.28 T.	823 d
		00 1.00 Ie=	.000 U
N	$r_{c} = \frac{17.274}{0018} + 0001486$	32 1 37 m -	799 4
			.105 u
P	$r_{c} = \frac{7.702}{.0001486}$	76 625 m r	012 4
	.0010 .0		.015 U
	СН		
Tube	(it.)	min lit.min	
0	$r_{c} = 15.53 .000106$ cal	c. found	960 4
	.00.14/ 1.4	55 1.55 ^r c ⁼	.005 u
N	$r_{c} = \frac{17.374}{000106}$	00 141 7 -	011 d
	.001147 1.5	1.41 Ic -	.011 U
P	$r_{c} = \frac{7.702}{000106}$	17 667 7	9654
	.001147 .7	13 .007 Ic=	.0000
	u		
Tube	n ₂ lit.r	nin.	
0	$r_{c_{-}}$ 15.53 .0000888 cal	c. found	culated for
	.000065 10.	hydrogen are b	eyond the
N	r_ <u>17.374</u> .0000888	rate limit of	the wet
	.000822 18.	any data with	this meter
P	r _{c = 7.702 .0000888}	for rates grea	ter than
	.0000832 8.	o or riters	per minute.



				Yee, Marine		Page 9
Tube	FLOW pressure cm. Hzo	DATA on CRI M E T E R temp.9C.	TICAL RA G A S temp.°C	TES. M E T E R . cu. ft.	TIME sec.	RATE LITERS per min. 760 mm. 25° C.
		AIR				
N	83.5	22.8		.1	49.6	3.3396
900 AIR	68.7	22.8		.1	55.1	3.060
	57.5	22.7		.1	58.5	2.880
	50.1	82.8		.1	60.4	3.808
	40.6	22.8		.1	64.7	2.504
	30.5	22.7		.1	79.4	2.124
	23.5	32.7	-	.1	101.2	1.663
	17.4	22.7	-4-5-5	.1	134.2	1.257
	7.7	22.6		.1	291.4	.579
0	85.3	22.9		. 1	61.6	2.736
AIR	75.0	33.0		.1	63.0	2.676
	68.3	23.0		.1	65.4	2.580
	49.6	23.0		.1	80.0	2.106
	35.2	23.9		.1	106.0	1.590
	34.4	22.9		.1	149.3	1.128
	18.0	32.8		.1	199.8	.824
	12.1	32.8		.1	290.0	.582
	81,2	22.7		.1	63.6	2.652
N 800 A IR	167.0	22.4	18	.1	31.9	5.350
	143.7	22.8	18	.1	34.9	4.99
	121.7	23.0	18	.1	38.4	4.54
	105.9	23.1	18	.1	41.6	4.14
	91.0	23.1	18	.1	45.3	3.84
	74.5	23.1	19	.1	50.5	3.44
	60.6	22.9	19	.1	55.4	3.14
	49.1	38.7	19	.1	59.4	3.93



Tube FLOW METER GAS METER TIME RATE liters pressure temp.°C temp. °C cu. ft. per minute sec. cm. H. 0 760 mm. 25° C AIR (continued) 22.3 38.0 19. 2.66 N .1 65.3 800 AIR 29.2 22.1 19. .]. 77.6 2.24 25.8 22.1 19. 2.01 .]. 86.3 19.0 22.0 19. .1 112.8 1.54 15.2 21.7 20. .1 137.6 1.26 8.0 55.0 20. . J. 247.8 .698 179.0 23.4 23. .1 40.2 4.18 0 40. 800 23.5 23. .1 43.9 3.84 AIR 153.0 47.2 3.57 135.0 23.6 23. .1 3.26 114.4 23.6 23. .1 51.6 56.6 3.98 .1 95.3 23.7 23. 0,0 2.75 61.4 .1 73.9 23.7 23. 2.57 65.7 23.7 23. .1 59.3 2.24 75.1 23. .1 23.7 47.2 1.83 92.2 .1 23. 37.4 23.7 1.5? 107.5 .1 23. 31.3 23.7 1.27 132.8 .]. 23. 24.8 23.7 196.0 .859 23. .) 16.2 23.7 .293 576.0 .1 5.4 23.7 23.

DATA on CRITICAL RATES (continued)


Tube	FLOW pressure cmH ₂ O	M E T E R temp.°C	G A S temp.°	METER C cu. ft.	TIME sec.	RATE, liters per minute 760 mm. 25°C
			AIR (co	ontinued)		
P 500 A IR	192.3 159.2 133.0 90.7 63.2 36.6 20.7	23.7 23.6 23.8 23.8 23.8 23.8 23.8 23.8 23.8	23 23 23 23 23 23 23 23 23	.1 .1 .1 .1 .1 .1	185.6 211.1 249.0 360. 504. 872.8 1527.2	.878 .800 .687 .469 .335 .194 .111
P 400	187.2 151.6 128.7 86.8 64.4 35.5 19.7	23.5 23.5 23.6 23.6 23.5 23.5 23.5 24.0	23 23 23 23 23 23 23 23 23	.1 .1 .1 .1 .1 .1	164.5 192.0 221.3 314.3 418. 748. 1346.	1.032 .875 .761 .535 .402 .225 .125

DATA on CRITICAL RATES (continued)



		DATA on C	RITICAL	RATES (cont	inued)	
Tube	FLOW pressure cmH ₂ O	METER temp.°C	GAS temp.°	METER C cu. ft.	TIME sec.	RATE, liters per minute 760 mm 25°C
		C	0 ₂			
N 900 CO ₂	84.5 74.7 66.9 56.1 45.3 36.8 27.8 18.1 13.7 9.3 7.0	23.1 23.2 23.2 23.2 23.1 23.1 23.1 23.1		.1 .1 .1 .1 .1 .1 .1 .1 .1	59.1 63.1 67.5 74.2 84.8 93.2 105.5 121.9 145.1 207.5 268.0	2.850 2.670 2.496 2.271 1.9884 1.8072 1.5966 1.3812 1.1598 .8119 .630
0 900 CO ₂	88.0 77.4 69 15 61.8 52.88 45.10 38.26 33.55 28.2 24.5 19.5 15.33 10.58 5.76	22.8 22.8 22.9 23.0 23.0 23.0 23.0 23.0 23.0 23.0 23.0	D1	.1 .1 .1 .1 .1 .1 .1 .1 .1 .1 .1 .1	77.5 84.0 89.2 95.0 102.8 111.0 121.3 127.0 135.3 162.2 200.2 286.2 497.0	2.178 2.101 1.890 1.788 1.644 1.518 1.428 1.392 1.326 1.2450 1.0398 .8430 .5898 .3396
N 800 CO _z	N3.8 8.8 13.1 15.0 26.6 43.4 63.4 84.7 112.5 142.9 172.3	C 0 22.2 22.3 24.0 23.6 24.3 24.7 25.0 25.2 25.2 25.4 25.6 25.7	2 21- 21- 21- 21- 21- 21- 22- 22-	.1 .1 .1 .1 .1 .1 .1 .1 .1	$\begin{array}{r} 421.4 \\ 198.8 \\ 141.8 \\ 127.6 \\ 101.7 \\ 80.4 \\ 65.1 \\ 54.9 \\ 46.6 \\ 40.4 \\ 36.3 \end{array}$	$\begin{array}{c} 0.402 \\ 0.852 \\ 1.192 \\ 1.325 \\ 1.663 \\ 2.106 \\ 2.59 \\ 3.073 \\ 3.618 \\ 4.17 \\ 4.645 \end{array}$



Tube	FLOW pressure cmH ₂ 0	METER temp.°C	G A S temp.°C	METER cu.ft.	TIME sec.	RATE, liters per minute 760 mm. 25°C
			COz			
0 300 CO ₂	178,0 142.9 102.0 65.5 44.4 29.1 21.7 15.8 10.8 0.5	25.9 26.0 26.1 26.1 26.2 26.3 26.5 26.4 26.5 26.2	88 88 88 88 88 88 88 88 88 88 88 88 88	.1 .1 .1 .1 .1 .1 .1 .1 .1	47.8 54.4 66.4 85.3 104.6 130.0 135.2 179.8 250.8 404.4	$\begin{array}{c} 3.525\\ 3.1\\ 2.54\\ 1.975\\ 1.61\\ 1.404\\ 1.243\\ 0.935\\ 0.76\\ 0.416\end{array}$
P 500 C0 ₂	197.8 148.3 115.6 72.5 36.8 52.1 32.7 90.7 170.6	25.1 24.9 24.4 24.2 23.9 24.0 23.8 23.0 23.1	C O₂ 25 25 25 25 25 25 25 25 25 25	.1 .1 .1 .1 .1 .1 .1 .1	227. 259. 288. 389. 721. 525. 1166. 334.8 239.4	.740 .649 .583 .432 .233 .320 .144 .501 .700

DATA on CRITICAL RATES (continued)



Tube	FLON pressure cmH ₂ 0	METER temp.°C	GASM temp.°C	E T E R cu. ft.	TIME sec.	RATE, liters per minute 760 mm 25°C
			C ₂ H ₄			
N 900 C ₂ H ₄	79.5 47.0 34.7 26.6 16.8 6.28 9.6 3.17 4.08 14.7 23.5 10.95	23.5 23.5 23.5 23.4 23.4 24.2 24.2 24.2 24.2 24.2 24.2	-	.1 .1 .1 .1 .1 .1 .1 .1 .1 .1 .1	47.6 64.0 76.0 86.2 105.0 203.0 140.0 388.0 302.0 106.8 92.0 127.6	3.516 2.616 2.202 1.944 1.596 .8250 1.197 .4332 1.5556 1.5714 1.842 1.314
N 800 C ₂ H ₄	145.5 131.3 115.0 93.8 79.45 61.8 44.8 34.2 26.45 17.35 11.45 7.0 3.9	21.8 22.0 22.2 22.3 22.3 22.4 22.3 22.3 22.3 22.2 22.2	C₂ Hy 22 22 22 22 22 22 22 22 22 22 22 5 22 5 22 5 23 23 23 23 23	.1 .1 .1 .1 .1 .1 .1 .1 .1 .1 .1	31.6 33.8 36.4 40.95 45.0 51.85 62.6 72.5 83.7 101.34 118.0 168.5 282.7	5.37 5.04 4.67 4.16 3.79 3.28 2.72 2.35 2.03 1.68 1.44 1.01 5.605
0 800 C _z H ₄	5.1 7.2 9.75 13.8 19.0 48.0 65.1 94.1 118.1 144.1 166.2	22.7 22.8 23.0 23.1 23.1 23.1 23.2 23.2 23.3 23.4 23.4	21 21 21 21 21 21 21 21 21 21	.1 .1 .1 .1 .1 .1 .1 .1 .1 .1 .1	347.1 246.5 193.9 141.4 120.0 81.2 68.9 55.5 48.6 43.2 39.7	0.492 0.692 0.88 1.21 1.42 2.1 2.48 3.08 3.505 3.95 4.29

DATA on CRITICAL BATES (continued)



		DATA on G	RITICAL HA	TES (cont:	inued)	
Tube	FLOW pressure cmH _z O	METER temp. °C	GAS M temp.°C	ETER cu.ft.	TIME sec.	RATE, liters per minute 760 mm 25°C
D		C2 H4	(continued	.)		
500 C _z H ₄	22.6 51.5 63.0 81.6 111. 148. 178.6	24.7 24.6 24.7 24.8 24.9 25.0 25.0	25 25 25 25 25 25 25	.1 .1 .1 .1 .1 .1	858.4 423.4 365.4 306. 245.8 217.0 194.2	.1955 .396 .459 .548 .672 .772 .863
P 400 C ₂ H ₄	188.1 161.7 138.4 113.6 82.6 60.4 42.7 20.0	24.4 24.7 24.9 25.0 25.2 25.2 25.2 25.4 25.4	24 24 24 24 24 25 25 25 25 25	.1 .1 .1 .1 .1 .1 .1	169.4 184.6 203.0 223.4 252.0 297. 389.6 769.8	.989 .909 .826 .750 .665 .565 .430 .218
			HYDROGEN			
N 800 H ₂	30.25 21.0 14.4 8.3	22.5 22.4 22.4 22.4 22.4	21 21 21 21	.1 .1 .1 .1	32.4 45.85 65.8 114.2	5.33 3.77 2.625 1.51
0 800 H ₂	67.2 55.4 41.7 29.4 18.5 8.9 51.0	21.2 21.2 21.3 21.3 21.3 21.3 21.4 21.7	20 20 20 20 20 20 20 20 20	.1 .1 .1 .1 .1 .1	23.2 27.75 36.55 51.15 81.10 166.1 29.9	7.43 6.22 4.72 3.37 2.132 1.035 5.78
P 400 H ₂	148.5 109.3 80.0 52.7	23.2 23.0 22.9 23.4	81 81 81 82 82	.1.5 .1 .1 .1	82.8 113.0 155.5 235.2	2.08 1.514 1.106 0.728



				11110 (00110	indou /	
Tube	FLOW pressure cmH ₂ O	METER temp.°C	GAS M temp.°C	ETER cu.ft.	TIME sec.	RATE, liters per minute 760 mm 25°C
		Н	YDROGEN			
P 100 H ₂	158.0 138.8 124.9 104.5 87.1 60.7 36.7 20.5 146.4 105.8 114.0	18.5 18.3 18.7 18.2 18.2 18.1 18.1 18.1 17.9 18.4 18.8	16 16 16 16 16 16 16 16 17 17 17	.1 .1 .1 .1 .1 .1 .1 .1 .1	27.6 30.4 33.15 37.83 43.85 59.55 92.7 157.3 28.9 37.6 35.20	6.25 5.68 5.21 4.56 3.936 2.893 1.861 1.095 5.94 4.57 4.87

DATA on CRITICAL RATES (continued)























END EFFECT

Having thus established the region where the break occurs, with a fair degree of accuracy, we decided to try to investigate the magnitude of end effect.

Poiseuille's law if plotted, rate against pressure, gives a straight line which is tangent to all of our curves at zero rate and zero pressure. Above this rate and pressure there is a deviation which increases with increase of pressure and rate. The rate of all gases is less than that calculated by Poiseuille's law. We will define "end effect" as the difference in pressure for a given rate, between that pressure actually found and the pressure which would correspond to this given rate if calculated by Poiseuille's law.

The value of the end effect can be found by combining with the pressure-rate diagram of a gas the plotted values calculated from Poiseuille's law and measuring the difference between the pressures which correspond to the same rate of flow.

Another angle of attack on this problem is to introduce into a tube an additional end effect and to measure the pressure difference at the same rates of flow between this tube and the original tube.

The method of introducing an added effect is best shown by the diagram on the following page.









Compound tube resulting from broken tube The tube was broken at right angles to its axis at point A. The broken ends were inserted into gas tight stoppers (b) which were in turn inserted into a short tube (c) whose diameter was very great compared to that of the capillary. A large diameter in this tube allows us to neglect the slight resistance arising

in it.

The gas could then flow through the tube as before, the only difference being that it encountered an additional end effect due to the extra ends. A pressure rate diagram from 1 this tube was then compared with a pressure-rate diagram of this same tube before it was broken and the magnitude of the increase in end effect measured from the two diagrams.

The data on these two methods of calculating "end effects" is given on the following pages.

The method of describing tubes is as follows; the diameter is indicated by a letter and the length in mm, by the figures following the letter If in the above illustration d-A was 800 mm. and A-c was 100mm the unbroken tube would be designated by a letter followed by 900; the compound tube would be the same letter followed by 800+ 100.



Tube	FLOW pressure cm. H ₂ O	METER temp.°C	GAS temp. AIR	METER °C cu.ft.	TIME sec.	RATE, liters PSE minute o
0 800 +100	67.7 56.4 43.7 36.1 17.0 24.4 19.3	23.1 23.1 23.1 23.1 23.1 23.1 23.5 23.5		.1 .1 .1 .1 .1 .1	70.4 80.0 98.0 113.2 220.5 159.4 198.0	2.383 2.094 1.712 1.481 0.761 1.050 0.846
N 800 +100	72.9 63.0 52.0 42.6 33.6 22.8 20.1 11.1 5.6	22.4 20.9 23.6 22.6 22.6 22.6 22.6 22.6 22.6 22.6	Air	•1 •1 •1 •1 •1 •1 •1 •1 •1	54.4 59.0 61.6 68.8 82.0 111.8 127.6 228.0 411.0	3.120 2.880 2.760 2.472 2.070 1.519 1.330 0.746 0.415

DATA on END EFFECT.



		DATA ON S	ND EFERGI	(continue	ea)	
Tube	FLOW pressure cm. Ho	METER temp.°C	C AS M E temp.°C	TEF cu.ft.	TIME sec.	RATE, litere per minute 760 mm 2 5 °C
N			COz			
800+1	162.5 177.5 137.3 105.6 124.0 150.7 10.1 13.3 16.3 20.2 25.6 33.8 45.2 57.7 71.8 83.6 41.5	23.8 23.8 23.8 23.8 23.8 23.8 23.8 23.1 22.1 22.1 22.1 22.1 22.2 22.2 22.2		.1 .1 .1 .1 .1 .1 .1 .1 .1 .1 .1 .1 .1	$\begin{array}{r} 42.4\\ 39.2\\ 46.2\\ 54.9\\ 50.0\\ 44.5\\ 215.6\\ 169.0\\ 145.3\\ 126.1\\ 114.5\\ 101.3\\ 86.4\\ 76.6\\ 67.3\\ 62.3\\ 58.4 \end{array}$	3.975 4.299 3.648 3.072 3.372 3.792 0.787 0.994 1.154 1.330 1.47 1.66 1.945 2.193 2.500 2.700 1.880
) 300+10	0		COz			
599.10	180.1 151.6 134.0 108.8 15.7 5.9 26.0 36.0 48.9 59.7 70.7 83.6 87.7 29.4 20.8 11.4 7.4	23.8 23.8 23.8 23.8 23.8 23.8 23.0 <t< td=""><td></td><td>.1 .1 .1 .1 .1 .1 .1 .1 .1 .1 .1 .1 .1</td><td>53.6 60.4 64.9 73.0 217.6 520.0 145.0 123.0 111.6 98.4 89.6 83.8 82.0 133.6 170.4 288.0 420.</td><td>3.1410 2.790 2.600 2.310 0.780 0.3258 1.167 1.374 1.516 1.719 1.876 2.019 2.063 1.266 0.992 0.5865 0.4020</td></t<>		.1 .1 .1 .1 .1 .1 .1 .1 .1 .1 .1 .1 .1	53.6 60.4 64.9 73.0 217.6 520.0 145.0 123.0 111.6 98.4 89.6 83.8 82.0 133.6 170.4 288.0 420.	3.1410 2.790 2.600 2.310 0.780 0.3258 1.167 1.374 1.516 1.719 1.876 2.019 2.063 1.266 0.992 0.5865 0.4020
N 600+2 H ₂	300 34.1 24.1 17.4 10.0	23.4 23.4 23.8 24.1	H g 23 23 23 23 23 23	.1 .1 .1 .1	29.7 41.6 56.2 95.3	5.725 4.08 1.025 1.785

ATA on END EFFECT (continued)











The curves plotted from the data on end effect indicate that errors in measurement, which are unimportant in so far as they affect the general shape of the curve itself, become serious obstacles when the same data is to be used in the calculation of end effect. These errors lead to results which differ among themselves when we attempt to derive a quantitative measure of end effect from the data. In order to come to any definite conclusion on the exact magnitude of end effect, the methods of measuring time, volume and pressure, particularly low pressures must be greatly refined.

Certain more or less qualitative conclusions may be drawn from the data that has been collected. Strictly speaking, these conclusions are applicable only to long tubes.

(1) For the gases used the end effect at a given rate increases as the diameter decreases, indicating that end effect may be a function of linear velocity.

(2) There are also indications that end effect is a function of the density of the gas. In all cases the end effect for hydrogen was very much less than the other gases, and in the great majority of cases the end effect of carbon dioxide was greater than that of air, which in turn , was greater than that of ethylene.

(3) The rate of change of end effect with change of rate of flow above the critical rate seems to be more nearly constant than below the critical rate. This gives rise to curves in this range that are so flattened as to approach straight lines in shape.


(4) No conclusions were drawn as to the relative magnitude of end effect for the various gases above the critical rate on account of insufficient data.

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(5) The formula for the critical rate, $r_c = 10 \text{ dv}$, was in one sense confirmed. This formula indicates that the critical rate is independent of the length of the tube.

In plotting our data on compound tubes it was found that the critical rates came at the same points, within the limit of error, that they occurred with the same tube before breaking and compounding. If the critical rate were dependent upon the length of the tube, it would seem that we would have found breaks occuring in regions other than those calculated on the basis that the tube was a continuus one whose length was equal to the sum of the lengths of the two segments. Some effect due to the shorter segment would surely have been noticed. (6) It is of interest to note that below the critical rate, the rate of flow of the gases used was in the inverse order of their viscosities. Above the critical rate howiver the order is the inverse of their densities. This is illustrated graph # 8.

Further work with these gases should prove, fruitful field of research provided that methods of taking measurements can be very greatly refined.











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SUMMARY

A set of experiments has been carried out on the rate of flow of air, carbon dioxide, ethylene and hydrogen, flowing through long capillary tubes.

The results of the experiments have been analized graphically by means of pressure-rate graphs. These graphs show that there is a critical rate for these gases and that at this critical rate their pressure-rate curve shows a decided change of slope.(This change in slope for hydrogen was not found because it lay outside of the range of our measurements) A formula giving the location of the break in the curve, proposed by Osborne Reynolds, has been found to give results that were in agreement with the experimental data.

An additional set of experiments was carried out to determine the magnitude of end effect. This was done by two different methods and the results obtained from the two methods were checked against each other.

The results of both sets of experiments show that the rate of flow a gas in a tube cannot be deduced by simple viscosity relations from the rate of flow of some other gas in the same tube, unless both the rate and pressure of the gases very nearly approach zero.







